

Reference: Van Cleve et. al. "Helium-3 Mining Aerostats in the Atmospheres of the Outer Planets", 2002



# Imagine an Interplanetary Future Where -



- d-He3 fusion produces most of Earth's energy needs without radioactivity or carbon emissions
- Space transportation has been revolutionized by an efficient fusion propulsion system with exhaust velocity up to 0.088 c
- Space commerce is stimulated by the existence of an interplanetary cargo worth \$3-M a kilogram
- Unmanned probes travel to the nearest star systems with flight times less than a human lifetime



## He-3 Fusion for Energy & Propulsion



- reactants are stable and storable
- products are energetic, charged and stable
  - Efficient electrical generation from MHD
  - No activation and embrittlement of reactor vessel
  - Efficient conversion to thrust with exhaust velocity up to 0.088 c --> ~50 yr interstellar flight using known physics.
- $3.6x10^{14}$  J/kg of d-He3 mixture =  $1.0x10^8$  kWh/kg
  - Fuel is about 20% of the kWh cost of electricity
  - If electricity is 15¢/kWh then He3 has a value of \$3M/kg

He-3 is one of the few commodities worth interplanetary freight costs



## Why Outer Planets for He-3?



- Earth: breeding of tritium from either isotope of lithium by neutron bombardment, tritons decay to He-3.
  - Containment, waste problems same as d-t fission.
  - USA has no current capability.
  - Lithium inventory?
- Moon: solar wind implanted in regolith, 10 ppb (10<sup>-8</sup>) by mass in uppermost few meters. ~1000 yr of 2001 energy needs- a starter catalyst?.
- Outer planets: primordial He3, ~10 parts per million (10<sup>-5</sup>), ~10<sup>9</sup> yr of 2001 energy needs- the ultimate energy source?.







#### Pro:

Closest to Earth and Sun

#### Con:

- Huge gravity means return vehicle has mass ratio >20 (nuclear thermal I<sub>sp</sub> = 900 s)
  - No mass budget left for cargo!
- A lot hotter at a any given density
  - Galileo probe killed by heat not by pressure



## Which Outer Planet-Saturn



### Pro:

- Not as far as Uranus and Neptune
- Rapid rotation substantially reduces ΔV to orbit Con:
- Seen as depleted ~5x in Helium compared to other outer planets
  - reanalysis of Voyager data 20 yr later restores that 5x- maybe
  - won't know for sure until we send an entry probe
- Rings as a navigation hazard
  - need close-in, co-orbiting mission to look





### Pro

- Primordial He3 abundance?
- $\Delta V$  to orbit requires mass ratio < 5
- Closer than Neptune

### Con

- Axial tilt complicates interplanetary travel
- Twice as far from Earth as Saturn

Uranus may be the closest planet without major possible problems -- but we must return to both Saturn to be sure

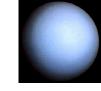


# Do we really know how much He3 is there?



- He3/He4 cannot be measured by remote sensing
- He3/H<sub>2</sub> and He3/He4 ratios have been measured in situ only by Galileo at Jupiter
- He3/He4 ratio of 10<sup>-4</sup> to 1.5x10<sup>-4</sup> from meteors, solar wind, cosmology
- Use Galileo results for He3/He4 = 10<sup>-4</sup> and Voyager results (?) for He4/H<sub>2</sub>





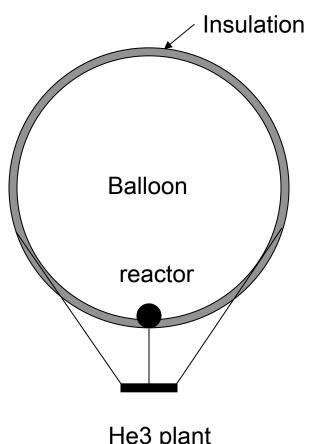
## He-3 Mining with Balloons

 Balloon diameter : 80 m

146 tonnes Total Plant mass:

Return vehicle: 59 tonnes

 Total lift needed: 205 tonnes



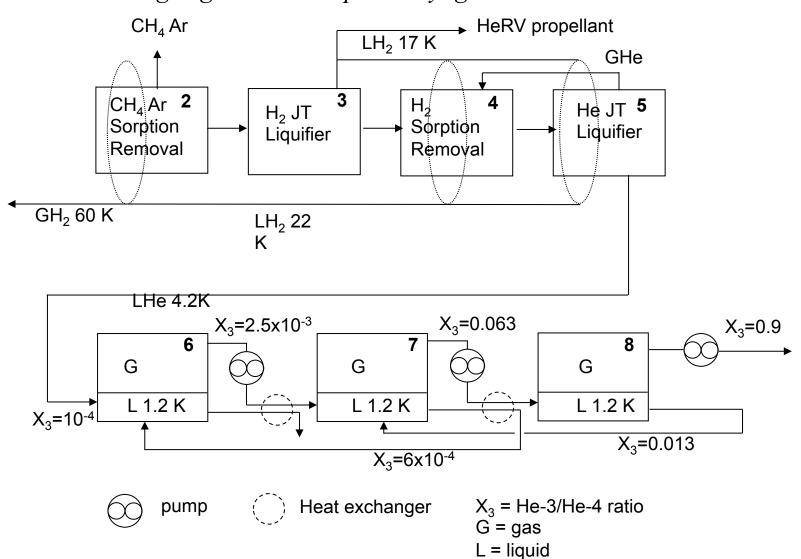
He3 plant



## Notional Distillation Plant Concept



Thinking Big about our Space Cryogenics Future





# Energy Economics $He3/H_2 = 10 ppm$



#### Table 1

Stage	Process	Energy (J)/g He3 7.2x10 <sup>7</sup>
1, 2, 3	cool atmosphere to 16 K	$7.2 \times 10^7$
3	liquify H <sub>2</sub> at 16 K	$3.2 \times 10^8$
5	cool He from 16 K to 4.2 K	$1.3 \times 10^7$
5	liquify He at 4.2 K	$1.1 \times 10^7$
6	cool LHe from 4.2 to 1.2 K	$1.2 \times 10^7$
total		$4.3x10^8$

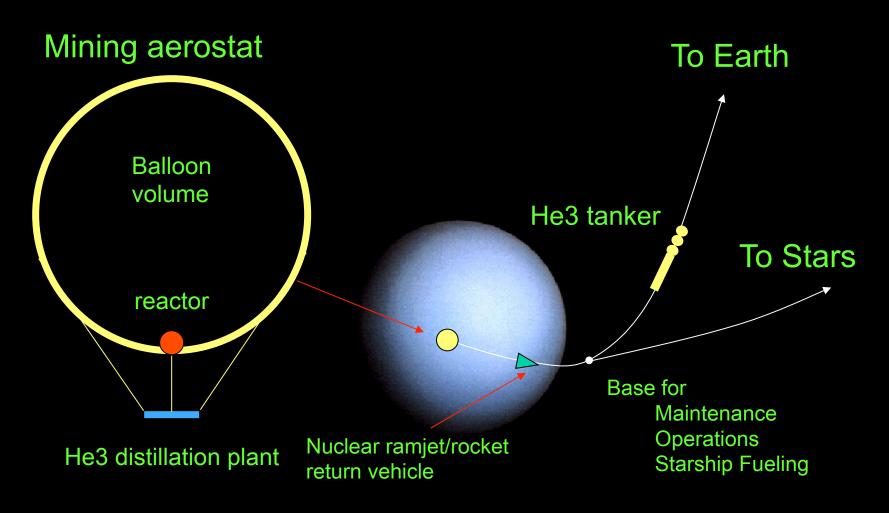
Transportation on 2 yr trajectory: 5x10<sup>7</sup>J/g He3

Energy released: 6x10<sup>11</sup> J/g He3

Theoretical energy payback: ~1000



### The Persian Gulf of the Solar System, 2150



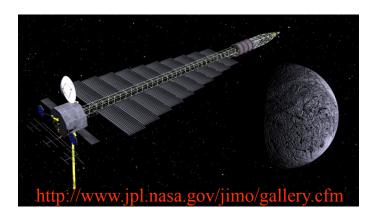
The most valuable interplanetary commodities are refined He-3, deuterium, and heavy metals



## Next Steps



- Jupiter Icy Moons Orbiter (JIMO)
  - nuclear fission-powered
  - electric propulsion flight system
  - Big deal: 20 tonnes, >\$4 B, 10 kWe
  - First of a series: Project Prometheus
- Saturn Ring Observer
- Uranus/Neptune Orbiter with Probes
- Self-deploying balloon probes for Mars, Titan
- Discovery/New Frontiers missions to other resource sites (Moon, asteroids, comets) for interplanetary commodity economy





### A Trial Balloon?



# Scientific balloon missions to outer planets, using Pu RTGs and/or O<sub>2</sub> burners, to study

- He3/He4 and He/H<sub>2</sub> ratio
- pressure vs. temperature for 1
- trace gas composition
- entry, deployment, and telemetry engineering experiments

A science balloon could be as small as 2.8 m diameter, and use at most 7 kg of Plutonium as a heat and power source